# SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPD&C - MAIN PROP. FMEA NO 05-6J -2171 -1 REV:06/14/88 ASSEMBLY :D & C PANEL C3A1 CRIT. FUNC: 1R P/N RI :ME452-0060- XXXX CRIT. HDW: 2 LEFT -4186 CENTER -4184 RIGHT -4185P/N VENDOR: VEHICLE 102 103 104 QUANTITY : 3 EFFECTIVITY: ¥ х :THREE PHASE(S): ₽L TO X 00 DO LS : REDUNDANCY SCREEN: A-PASS B-FAIL C-PASS PREPARED BY: APPROVED BY: APPROVED BY (NASA): *1*₩₩ J BROWN DES DES EPDC SSM Counterly provided the Charles

6/27/38

SSK.

EPDC RESAWAJ J

22/186

MPS

MPS

тичем.

REL

QΕ

PUSHBUTTON SWITCH, (TWO POLES), MAIN ENGINE SHUTDOWN CIRCUIT.

20, January 1 27 28

#### FUNCTION:

PROVIDES MANUAL MAIN ENGINE SHUTDOWN CAPABILITY. 35V73A3A1S12, 13, 14.

REL

0E

# PATLURE MODE:

FAILS OPEN.

DEFENSOR

TOP~ D MASAI

# CAUSE(S):

PIECE PART STRUCTURAL FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK, PROCESSING ANOMALY.

#### EFFECT(S) ON:

- (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE (E) FUNCTIONAL CRITICALITY
- (A) LOSS OF COMMANDS TO SWITCH REDUNDANCY MANAGEMENT (RM).
- (B) LOSS OF MANUAL SHUTDOWN AND SAFING COMMAND CAPABILITY.
- (C,D) NO EFFECT FIRST FAILURE.

# SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPO&C - MAIN PROP. FMEA NO 05-6J -2171 -1 REV: 06/14/88

- (E) 1R/2, 1 SUCCESS PATH AFTER FIRST FAILURE. TIME FRAME ASCENT.

   SWITCH FAILS OPEN, RESULTING IN LOSS OF MANUAL SHUTDOWN AND SAFING COMMAND CAPABILITY.
  - 2) ASSOCIATED SSME EXPERIENCES UNCONTAINED SHUTDOWN, INCLUDING DAMAGE WHICH CAUSES LOSS OF DATA PATH.

FOR ENGINE-OUT FAILURE INCLUDING LOSS OF DATA PATH, CREW USES SHUTDOWN PUSHBUTTON SWITCH TO INITIATE PREVALVE CLOSURE SEQUENCE WHICH ISOLATES PROPELLANT SUPPLY. FIRST FAILURE RESULTS IN LOSS OF THIS CAPABILITY. SECOND FAILURE THEN RESULTS IN PROPELLANT LEAKAGE INTO THE AFT COMPARTMENT, OVERPRESSURIZATION, AND FIRE/EXPLOSION HAZARD. POSSIBLE LOSS OF CRITICAL ADJACENT FUNCTIONS DUE TO CRYOGENIC EXPOSURE. POSSIBLE LOSS OF CREW/VEHICLE.

FAILS B SCREEN BECAUSE NO INSTRUMENTATION IS AVAILABLE TO DETECT FAILURE.

#### DISPOSITION & RATIONALE:

- (A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE
- (A-D) FOR DISPOSITION AND RATIONALE REFER TO APPENDIX A, ITEM NO. 3 - SWITCH, PUSHBUTTON.
- (B) GROUND TURNAROUND TEST
  SSME SHUTDOWN SWITCH VERIFICATION V41AFO.270 EVERY FLIGHT.
- (E) OPERATIONAL USE
  NO CREW ACTION CAN BE TAKEN.